THE PAPUA NEW GUINEA UNIVERSITY OF TECHNOLOGY

MECHANICAL ENGINEERING – 3rd YEAR DEGREE

FIRST SEMESTER EXAMINATIONS – 2022

ME 113- ENGINEERING MATERIALS & PROPERTIES

MONDAY, 6TH JUNE 2022 – 12.50 PM

TIME ALLOWED: 3 HOURS

INSTRUCTIONS FOR CANDIDATES

- 1. You have 10 minutes to read the paper. You **must not** begin writing during this time.
- 2. Answer any four (4) questions and you can do them in any order.
- 3. **Use only ink.** Do not use pencil for writing except for drawings and sketches.
- 4. Start each question on a new page and show all your calculations in the answer book provided. No other written material will be accepted.
- 5. Write your **NAME** and **NUMBER** clearly on the front page. **Do it now.**
- 6. Calculators are permitted in the examination room. Notes, textbooks or smart phones are not allowed.

MARKING SCHEME:

Question Number 1	15 Marks
Question Number 2	15 Marks
Question Number 3	15 Marks
Question Number 4	15 Marks
Question Number 5	15 Marks

Question 1

a) Briefly describe the following types of bonding:

(i) Ionic Bonding (2.5 Marks)

(i). Metallic bonding (2.5 Marks)

b) The force of attraction between a divalent cation and a divalent anion is $1.67 \times 10^{-8} \text{ N}$. If the ionic radius of the cation is 0.080 nm, what is the anion radius?

Question 2

a) What is the difference between atomic structure and crystal structure? (2.5 Marks)

b) Define atomic packing factor (APF) (2.5 Marks)

c) Titanium (Ti) has an HCP crystal structure and a density of 4.51 g/cm^3 . (a) What is the volume of its unit cell in cubic meters? (b) If the c/a ratio is 1.58, compute the values of c and a.

(10 Marks)

Question 3

a) Describe the following terms;

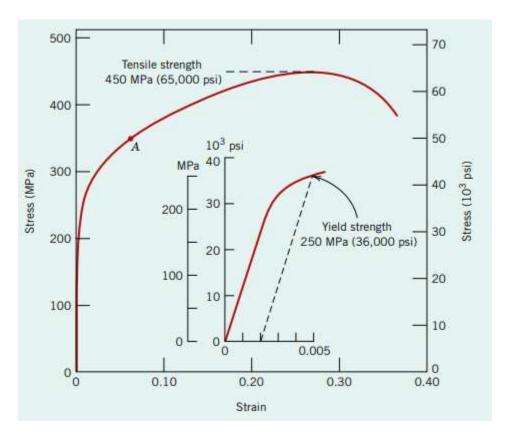
(i) Tensile strength (2.5 Marks)

(ii) Hardness (2.5 Marks)

b) Consider the brass alloy for which the stress–strain behavior is shown below. A cylindrical specimen of this material 10.0 mm in diameter and 101.6 mm long is pulled in tension with a force of 10,000 N (2250 lbf). If it is known that this alloy has a value for Poisson's ratio of 0.35, compute

(i) the specimen elongation (5 Marks)

(ii) the reduction in specimen diameter (2.5 Marks)



Question 4

a) Briefly describe the following terms;

(i) Brittle fracture (2.5 Marks) (ii) Ductile fracture (2.5 Marks)

- b) The fatigue data for a steel alloy are given as follows:
- (i) Make an S–N plot (stress amplitude versus logarithm of cycles to failure) using these data.

(2.5 *Marks*)

- (ii) What is the fatigue limit for this alloy? (2.5 Marks)
- (iii) Determine fatigue lifetimes at stress amplitudes of 415 MPa and 275 MPa. (2.5 Marks)
- (iv) Estimate fatigue strengths at 2×10^4 and 6×10^5 cycles. (2.5 Marks)

Question 5

a) Briefly describe the following terms;

(i) Fatigue (endurance) limit (2.5 Marks)

(ii) Fatigue strength (2.5 Marks)

b) A fatigue test was conducted in which the mean stress was 70 MPa, and the stress amplitude was 210 MPa.

(i) Compute the maximum and minimum stress levels. (3.5 Marks)

(ii) Compute the stress ratio. (3.5 Marks)

(iii) Compute the magnitude of the stress range. (3 Marks)

$$F_A = \frac{dE_A}{dr} = \frac{d\left(-\frac{A}{r}\right)}{dr} = -\left(\frac{-A}{r^2}\right) = \frac{A}{r^2}$$
 (2.12)

$$F_A = \frac{1}{4\pi\epsilon_0 r^2} (|Z_1|e)(|Z_2|e) \tag{2.13}$$

$$F_A = \frac{1}{4\pi (8.85 \times 10^{-12} \,\mathrm{F/m})(r^2)} [|Z_1|(1.602 \times 10^{-19} \,\mathrm{C})][|Z_2|(1.602 \times 10^{-19} \,\mathrm{C})]$$

$$= \frac{(2.31 \times 10^{-28} \,\mathrm{N \cdot m^2})(|Z_1|)(|Z_2|)}{r^2}$$
(2.14)

$$r_0 = r_{K^+} + r_{Br^-} (2.15)$$

3.4	$a = \frac{4R}{\sqrt{3}}$	Unit cell edge length, BCC	56
3.8	$ ho = rac{nA}{V_C N_{ m A}}$	Theoretical density of a metal	60

Equation Summary

Equation Number	Equation	Solving For	Page Number
6.1	$\sigma = rac{F}{A_0}$	Engineering stress	172
6.2	$\epsilon = rac{l_i - l_0}{l_0} = rac{\Delta l}{l_0}$	Engineering strain	172
6.5	$\sigma = E\epsilon$	Modulus of elasticity (Hooke's law)	174
6.8	$v = -\frac{\epsilon_x}{\epsilon_z} = -\frac{\epsilon_y}{\epsilon_z}$	Poisson's ratio	177
6.11	$\%EL = \left(\frac{l_f - l_0}{l_0}\right) \times 100$	Ductility, percent elongation	184
6.12	$%RA = \left(\frac{A_0 - A_f}{A_0}\right) \times 100$	Ductility, percent reduction in area	184

Equation Summary

Equation Number	Equation	Solving For	Page Number
8.1	$\sigma_m = 2\sigma_0 \left(\frac{a}{\rho_t}\right)^{1/2}$	Maximum stress at tip of elliptically shaped crack	258
8.4	$K_c = Y\sigma_c\sqrt{\pi a}$	Fracture toughness	260
8.5	$K_{lc} = Y\sigma\sqrt{\pi a}$	Plane-strain fracture toughness	261
8.6	$\sigma_{\!\scriptscriptstyle c} = rac{K_{Ic}}{Y \sqrt{\pi a}}$	Design (or critical) stress	262
8.7	$a_c = rac{1}{\pi} \left(rac{K_{Ic}}{\sigma Y} ight)^2$	Maximum allowable flaw size	263
8.14	$\sigma_m = \frac{\sigma_{\max} + \sigma_{\min}}{2}$	Mean stress (fatigue tests)	270
8.15	$\sigma_r = \sigma_{\text{max}} - \sigma_{\text{min}}$	Range of stress (fatigue tests)	270
8.16	$\sigma_a = \frac{\sigma_{\max} - \sigma_{\min}}{2}$	Stress amplitude (fatigue tests)	270
8.17	$R = rac{\sigma_{\min}}{\sigma_{\max}}$	Stress ratio (fatigue tests)	270
8.23	$\sigma = \alpha_l E \Delta T$	Thermal stress	281
8.24	$\dot{\epsilon}_s = K_1 \sigma^n$	Steady-state creep rate (constant temperature)	283
8.25	$\dot{\epsilon}_s = K_2 \sigma^n \exp\left(-\frac{Q_c}{RT}\right)$	Steady-state creep rate	284
8.27	$m = T(C + \log t_r)$	Larson-Miller parameter	285